

# ENGR 244 (Mechanics of Materials)

## Lab No. 1: BRINELL HARDNESS TEST

### DATA AND CALCULATION SHEET

Student Name: \_\_\_\_\_ Student ID: \_\_\_\_\_ Lab Group: \_\_\_\_\_

SAMPLE	No. of Trial	Ball Diameter, D (mm)	Diameter of impression, d (mm)			Load, P		Depth of indentation, t (mm)	Area of indentation, A (mm <sup>2</sup> )	Brinell Hardness, HB (Kg/mm <sup>2</sup> )	Average Brinell Hardness HB (Kg/mm <sup>2</sup> )
			d <sub>x</sub>	d <sub>y</sub>	d = ½(d <sub>x</sub> +d <sub>y</sub> )	in Newton (N)	in Kg (N/9.81)				
Steel	1	10.00									
	2										
	3										
Aluminum	1	10.00									
	2										
	3										

Formula used

$$HB = \frac{P(kg)}{A(mm^2)} \text{ and } A = \pi Dt = \frac{\pi D (D - \sqrt{D^2 - d^2})}{2}$$

where:

P = applied load in kg (Load (N)/g).  
 g = acceleration due to gravity (9.81 m/s<sup>2</sup>)  
 A = area of the indentation (mm<sup>2</sup>)

D = diameter of the ball (mm)  
 d = mean diameter of the impression (mm)  
 t = depth of indentation (mm)

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## Lab No. 2: TENSION TEST ON METALS

# DATA SHEET

Student Name: \_\_\_\_\_ Student ID \_\_\_\_\_ Lab Group \_\_\_\_\_

Sample	Before Fracture	After Fracture
Steel	$L_i = \dots\dots\dots$ mm, $d_i = \dots\dots\dots$ mm	$L_f = \dots\dots\dots$ mm, $d_f = \dots\dots\dots$ mm
Aluminum	$L_i = \dots\dots\dots$ mm, $d_i = \dots\dots\dots$ mm	$L_f = \dots\dots\dots$ mm, $d_f = \dots\dots\dots$ mm

ALUMINUM				STEEL			
Load, P(N)	Deformation, $\delta$ (mm)	Load, P(N)	Deformation, $\delta$ (mm)	Load, P (N)	Deformation, $\delta$ (mm)	Load, P (N)	Deformation, $\delta$ (mm)
Maximum Load, $P_{max} = \dots\dots\dots$ N				Maximum Load, $P_{max} = \dots\dots\dots$ N			

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## Lab No. 3: TORSION TEST ON METALS

### DATA AND CALCULATION SHEET

Student Name: \_\_\_\_\_ Student ID: \_\_\_\_\_ Lab Group: \_\_\_\_\_

Sample	Diameter, $D = (D_1 + D_2 + D_3)/3$	Gauge Length, $L$	Polar Moment of inertia, $J$
Steel	mm	mm	$\text{mm}^4$
Aluminum	mm	mm	$\text{mm}^4$
Brass	mm	mm	$\text{mm}^4$

Steel				
T (N.m)	θ (Deg.)	θ (Rad.)	G (GPa)	G <sub>av</sub> (GPa)
1.50				
3.00				
4.50				
6.00				
7.50				
9.00				
10.50				

Aluminum				
T (N.m)	θ (Deg.)	θ (Rad.)	G (GPa)	G <sub>av</sub> (GPa)
1.00				
2.00				
3.00				
4.00				
5.00				
6.00				

Brass				
T (N.m)	θ (Deg.)	θ (Rad.)	G (GPa)	G <sub>av</sub> (GPa)
1.00				
2.00				
3.00				
4.00				
5.00				
6.00				

Formulas

$J = (\pi D^4)/32$       where T = Torque (N.mm), L= Gauge length (mm), D = Diameter of the specimen (mm)  
 $G = TL/ \theta J$       θ = angle of twist (radians), J= Polar Moment of inertia ( $\text{mm}^4$ )

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## Lab No. 4: STRESS ANALYSIS OF BEAMS USING STRAIN GAUGES

Student Name:

Student ID:

Lab Group:

Properties of the steel beam:

Span length between the two loading points:

Span length between supports,  $L = 455$  mm,

$2c =$  \_\_\_\_\_ mm.

Width,  $b =$  \_\_\_\_\_ mm,

Height,  $h =$  \_\_\_\_\_ mm

Applied Load, P (N)	Strain Indicator reading units are in micro-strain ( $10^{-6}$ strain)					
	Channel # 1	Channel # 2	Channel # 3	Channel # 4	Channel # 5	Channel # 6
1000						
2000						
3000						
4000						
5000						
Unload						

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## Lab No. 5: DEFLECTION OF BEAMS

### DATA SHEET

Student Name: \_\_\_\_\_ Student ID: \_\_\_\_\_ Lab Group: \_\_\_\_\_

Dimensions	BRASS	STEEL	ALUMINUM
Span Length, L (mm)	455	455	455
Width, w (mm)			
Height, h (mm)			

DEFLECTION OF BEAM, $\delta$ (mm)						
Applied Load	BRASS		STEEL		ALUMINUM	
	at $x = L/2$	at $x = L/4$	at $x = L/2$	at $x = L/4$	at $x = L/2$	at $x = L/4$
200 N						
400 N						
600 N						
800 N						
1000 N						

### PART 2 : CANTILEVER BEAMS

Brass width x height .....X.....(mm), Steel width x height .....X.....(mm), Alum width x height.....X.....(mm)

CANTILEVER BEAM DEFLEXION, $\delta$ (mm)						
Applied Load	BRASS span L=.....mm		STEEL span L=.....mm		ALUM span L=.....mm	
	$x=L/2$	$x=L$	$x=L/2$	$x=L$	$x=L/2$	$x=L$
100 gm						
200 gm						
300 gm						
400 gm						
500 gm						

Analysis & Discussion: Follow the instructions given on page # 21 of your lab manual.

## EXPERIMENT 6: Typical DATA Sheet

Type of column	Column Length, L (mm)	End Condition	Effective length factor, k	Effective length, $L_{\text{eff}}$ (mm)	Inner diameter, $d_{\text{in}}$ (mm)	Outer diameter, $d_{\text{out}}$ (mm)	Cross sectional area, A ( $\text{mm}^2$ )	Moment of inertia, I ( $\text{mm}^4$ )	Radius of gyration, r (mm)	Slenderness ratio, $\lambda$	Allowable Stress, $\sigma_{\text{all}}$ (MPa)	Theoretical Critical load, $P_{\text{cr}}$ (N)	Theoretical Critical stress, $\sigma_{\text{cr}}$ (MPa)	Experimental Critical Load, $P_{\text{cr}}$ (N)	Experimental Critical stress, $\sigma_{\text{cr}}$ (MPa)
<b>Hollow</b>	75	Pin-Pin	1	75											
	125	Pin-Pin	1	125											
	225	Pin-Pin	1	2:25											
	225	Pin-Fixed	0.707												
	225	Fixed-Fixed	0.5												
<b>Solid</b>	225	Pin-Pin	1	225											
	225	Pin-Fixed	0,707												
	225	Fixed-Fixed	0.5												

### Report:

1. Tabulate the critical loads measured for various specimens
2. On ONE graph, show the comparison of the followings:
  - (i) Theoretical  $\sigma_{\text{cr}}$  vs.  $\lambda$ ,
  - (ii) Experimental  $\sigma_{\text{cr}}$  vs.  $\lambda$ , and
  - (iii) Allowable stress vs.  $\lambda$ ,

Discussion: Answer all FOUR questions given in your Lab Manual.

### Effective Length factor, k

One end fixed, one end free	Both ends pinned	One end fixed, one end pinned	Both ends fixed
$K = 2$	$K = 1$	$K = \frac{1}{\sqrt{2}}$	$K = \frac{1}{2}$

(Sometimes 'Fixed' is called 'clamped', 'pinned' is called 'hinged')