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### CHG 2312 – Fluid Flow Assignment #2 Solution

#### Question 1:

Given:

$$D = 0.2 \text{ m}$$

$$V = 6 \text{ m/s}$$

$$v = U \cdot [2 - (x)^{1/2}] / 4$$

Find: Discharge velocity,  $U$

Assumptions: Steady State, Incompressible Fluid

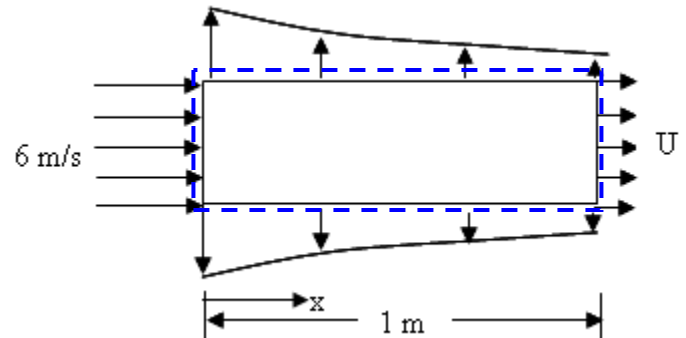
Solution:

First define a control volume.

Second define the inlet and outlet of the system.

Inlet: left side of the pipe where velocity is at a constant value of 6 m/s.

Outlet: body of the pipe (with variable velocity) and end of the pipe on the right-hand side (with velocity of  $U$ ).



Mass balance:

$$\frac{dm_{C.V.}}{dt} = \dot{m}_{in} - \dot{m}_{out} = 0$$

$$\dot{\rho}Q_{in} = \dot{\rho}Q_{out,body} + \dot{\rho}Q_{out,end}$$

$$V_{in} A_{in} = v A_{out,body} + U A_{out,end}$$

$$\left(\frac{\pi D^2}{4}\right) V_{in} = \int_0^1 \left[ (\pi D dx) \left( U \cdot \frac{2 - x^{1/2}}{4} \right) \right] + U \left( \frac{\pi D^2}{4} \right)$$

$$DV_{in} = U \left( \int_0^1 [2 - x^{1/2}] dx \right) + DU$$

$$DV_{in} = U \left[ 2x - \frac{2x^{3/2}}{3} \right]_0^1 + DU$$

$$U = \frac{DV_{in}}{D + \left[ 2x - \frac{2x^{3/2}}{3} \right]} = \frac{(0.2m)(6m/s)}{(0.2m) + [2 - 2/3]} = 0.78m/s$$



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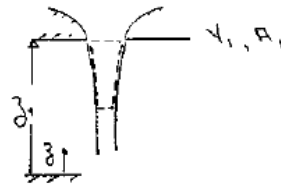
#### Question 2:

##### Given:

- 1- Liquid stream leaving a nozzle pointing downward as shown
- 2- Frictionless flow
- 3- Uniform flow

##### Assumption:

- 1- Steady flow
- 2- Incompressible flow
- 3- Flow along a streamline
- 4-  $P = P_1 = P_{atm}$



**Find:** Variation in jet area for z,  $z_0$

##### Solution:

From the Bernoulli equation

$$\Delta \left( \frac{P}{\rho} + gz + \frac{V^2}{2} \right) - \frac{dW_{n.s.f}}{dm} = 0$$

$$\Rightarrow V^2 = V_1^2 + 2g(z_1 - z)$$

And

$$V_1 A_1 = VA$$

Thus

$$V_1^2 \left( \frac{A_1}{A} \right)^2 = V_1^2 + 2g(z_1 - z)$$

Solving for A

$$A = A_1 \sqrt{\frac{1}{1 + \frac{2g(z_1 - z)}{V_1^2}}}$$

Note: Jet area decreases as z decreases, owing to the higher velocity. You can observe such behaviour from any water tap in kitchens or bathrooms.

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**Question 3:**

Given:

$$D_1 = 0.1 \text{ m}$$

$$D_2 = 0.05 \text{ m}$$

$$P_2 = P_{\text{atm}}$$

$$V_2 = 20 \text{ m/s}$$

$$Z_1 = 0 \text{ m}$$

$$Z_2 = 4 \text{ m}$$

Find:

- 1) Gauge Pressure,  $P_1$
- 2) Gauge Pressure,  $P_1$ , if device was inverted

Assumptions:

- 1) Steady Flow
- 2) Incompressible Fluid
- 3) Frictionless Flow
- 4)  $P_2$  gauge = 0
- 5)  $z_1 = 0 \text{ m}$
- 6) no external work

Solution:

Apply continuity to CV shown to determine  $V_1$ ; the Bernoulli equation is then applied along a streamline from 1 to 2 to determine  $P_1$ .

Basic Equations:

$$\frac{dm_{cv}}{dt} = \dot{m}_{in} - \dot{m}_{out} = 0 \quad (\text{Continuity})$$

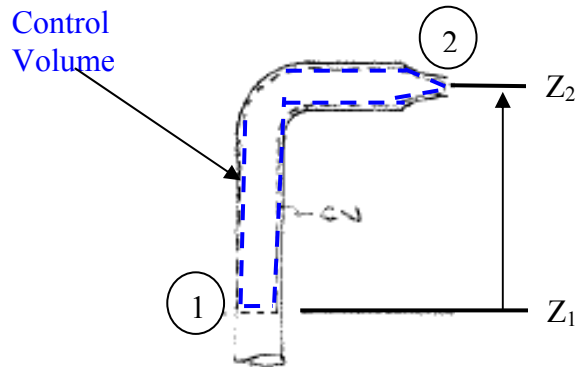
$$\frac{P_2 - P_1}{\rho} + \frac{V_2^2 - V_1^2}{2} + g(\Delta z) = \frac{dW}{dm} - F \quad (\text{Bernoulli})$$

From the continuity equation:

$$\rho_1 V_1 A_1 = \rho_2 V_2 A_2 \Rightarrow V_1 A_1 = V_2 A_2 \text{ or } V_1 = \left(\frac{D_2}{D_1}\right)^2 \cdot V_2$$

From the Bernoulli equation:

$$P_1 = \rho \left( \frac{V_2^2 - V_1^2}{2} + gz_2 \right) = \rho \left( \frac{V_2^2}{2} \left( 1 - \frac{V_1^2}{V_2^2} \right) + gz_2 \right) = \rho \left( \frac{V_2^2}{2} \left( 1 - \left( \frac{D_2}{D_1} \right)^4 \right) + gz_2 \right)$$





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$$P_1 = \left( 999 \frac{\text{kg}}{\text{m}^3} \right) \left( \frac{(20 \frac{\text{m}}{\text{s}})^2}{2} \left( 1 - \left( \frac{0.05\text{m}}{0.1\text{m}} \right)^4 \right) + 9.81 \frac{\text{m}}{\text{s}^2} \cdot 4\text{m} \right) = 227 \frac{\text{kN}}{\text{m}^2} = 227 \text{kPa (gauge)}$$

If the device is inverted,  $z_2 = -4$  m with  $z_1 = 0$  m

$$P_1 = 999 \frac{\text{kg}}{\text{m}^3} \left( \frac{(20 \frac{\text{m}}{\text{s}})^2}{2} \left( 1 - \left( \frac{0.05\text{m}}{0.1\text{m}} \right)^4 \right) + (9.81 \frac{\text{m}}{\text{s}^2}) \cdot (-4\text{m}) \right) = 148 \frac{\text{kN}}{\text{m}^2} = 148 \text{kPa (gauge)}$$



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### CHG 2312 – Fluid Flow Assignment #2 Solution

#### Question 4:

$$\rho_w = 999 \text{ kg/m}^3, \mu_w = 0.98 \times 10^{-3} \text{ Pa}\cdot\text{s}$$

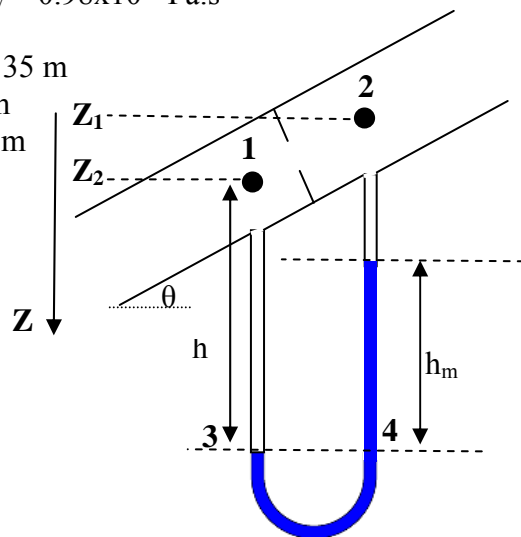
$$\rho_m = 13600 \text{ kg/m}^3$$

$$h_m = 135 \text{ mm} = 0.135 \text{ m}$$

$$D_2 = 6 \text{ cm} = 0.06 \text{ m}$$

$$D_1 = 10 \text{ cm} = 0.10 \text{ m}$$

$$\theta = 30^\circ$$



Assumptions:

Steady-state flow, Incompressible fluid, inviscid fluid, **Point 2 is very close to the orifice centre**

Mass balance from point 1 to 2:

$$\frac{dm_{c.v.}}{dt} = \dot{m}_{in} - \dot{m}_{out} = 0$$

$$\rho V_1 A_1 = \rho V_2 A_2 \quad \Rightarrow \quad V_1 = \frac{A_2}{A_1} V_2 = \frac{D_2^2}{D_1^2} V_2$$

Energy balance, B.E.:

$$\Delta \left( \frac{P}{\rho} + gz + \frac{V^2}{2} \right) - W + F = 0 \quad \text{where } W = 0, F = 0$$

$$\Delta \left( \frac{P}{\rho} + gz + \frac{V^2}{2} \right) = 0$$

$$\frac{P_2 - P_1}{\rho} + g(z_2 - z_1) + \frac{V_2^2 - V_1^2}{2} = 0$$

$$\frac{P_2 - P_1}{\rho} + g(z_2 - z_1) + \frac{V_2^2}{2} \left( 1 - \frac{D_2^4}{D_1^4} \right) = 0$$

$$V_2^2 = 2 \left( \frac{P_2 - P_1}{\rho} + g(z_2 - z_1) \right) / \left( 1 - \frac{D_2^4}{D_1^4} \right)$$

$$V_2 = \left( \frac{2 \frac{P_2 - P_1}{\rho} + 2g(z_2 - z_1)}{\left( 1 - \frac{D_2^4}{D_1^4} \right)} \right)^{1/2}$$



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Determine  $P_1 - P_2$  from manometer reading:

$$P_1 + \rho_w g h - \rho_m g h_m - \rho_w g (h - h_m + (z_2 - z_1)) - P_2 = 0$$

$$P_1 - P_2 = \rho_m g h_m - \rho_w g h_m + \rho_w g (z_2 - z_1)$$

Mass flow rate in the orifice is:

$$\dot{m} = \rho V_2 A_2 = \rho \pi \frac{D_2^2}{4} C_v \left( \frac{2 \frac{P_1 - P_2}{\rho} + 2g(z_1 - z_2)}{\left(1 - \frac{D_2^4}{D_1^4}\right)} \right)^{1/2} = \rho \pi \frac{D_2^2}{4} C_v \left( \frac{2\rho_m g h_m - 2\rho g h_m + 2\rho g (z_2 - z_1) + 2g(z_1 - z_2)}{\rho \left(1 - \frac{D_2^4}{D_1^4}\right)} \right)^{1/2}$$

$$\dot{m} = C_v \rho \pi \frac{D_2^2}{4} \left( \frac{2g h_m (\rho_m - \rho)}{\rho \left(1 - \frac{D_2^4}{D_1^4}\right)} \right)^{1/2}$$

Determine  $C_v$ :

1. find  $V_2$  without  $C_v$

$$V_2 = \left( \frac{2\rho_m g h_m - 2\rho g h_m}{\rho \left(1 - \frac{D_2^4}{D_1^4}\right)} \right)^{1/2} = \left( \frac{2(9.8)(0.135m)(13600 - 999)}{(999)\left(1 - \left(\frac{0.06}{0.1}\right)^4\right)} \right)^{1/2} = 6.2 \text{ m/s}$$

2. find Reynolds:

$$Re = \frac{\rho D_2 V_2}{\mu} = \frac{(999)(0.06)(6.2)}{(0.89 \times 10^{-3})} = 417559$$

3. read  $C_v$  from graph

$$C_v = 0.61$$

4. Calculate new  $V_2$  by including  $C_v$ , and then new Re number:

$$V_2 = 3.8 \text{ m/s}$$

$$Re = 243249$$

$$\text{New } C_v = 0.61$$

**Since new  $C_v$  and old  $C_v$  are similar then no more calculation is required and  $V_2 = 3.8 \text{ m/s}$**

Determine mass flow rate:

$$\dot{m} = \rho \pi \frac{D_2^2}{4} V_2 = (999) \pi (0.06)^2 (3.8) / 4 = 10.73 \text{ kg/s}$$



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## CHG 2312 – Fluid Flow Assignment #2 Solution

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### Question 5

#### Given:

Fluid: Air

$D_2 = 3$  inch

$D_1 = 6$  inch

$P_1 = 60$  psi

$T_1 = T_2 = 68$  °F = 20°C

#### Assumptions:

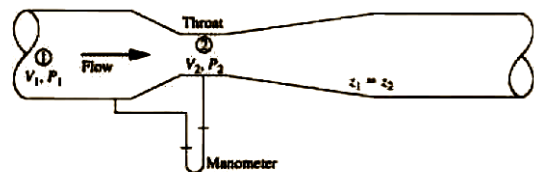
- 1- Density change is negligible
- 2- Ideal gas
- 3- Steady state system
- 4- Frictionless Flow

Find: Maximum flow rate for incompressible flow; Pressure reading

#### Solution:

In order to determine the mass flow rate of the air in the venturi meter, we could apply the B.E. equation. However, for any gas to be counted as an incompressible flow (important criteria for using B.E. equation), its velocity has to be low. In your text book this limit has been set at 200 ft/s = 63 m/s, whereas other literature use velocity of 100 m/s. The later number has been used here.

Looking at a venturi meter arrangement and again for applying B.E., then the maximum gas velocity in the system will be 100 m/s. But now the question would be where would the gas has a maximum velocity (i.e., at upstream of piping or at the throat).



As we know the highest velocity will be at the throat of the venturi since it has the smallest diameter. Therefore,  $V_2 = 100$  m/s. Therefore, we can easily calculate the maximum mass flow rate as:



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$$\dot{m} = \rho V_2 A_2$$

$$\rho = \frac{PM}{RT} = \frac{(60 \text{ psi} \frac{101325 \text{ Pa}}{14.7 \text{ psi}})(29 \text{ kg / kmol})}{(8.314 \times 10^3 \frac{\text{m}^3 \text{ Pa}}{\text{K.kmol}})(20 + 273) \text{ K}} = 4.9 \text{ kg / m}^3$$

$$\dot{m} = (4.9 \text{ kg / m}^3)(100 \text{ m / s})(\pi \times (3 \text{ inch} \frac{1 \text{ m}}{39.37 \text{ inch}})^2 / 4) = 2.23 \frac{\text{kg}}{\text{s}}$$

The pressure drop (i.e. reading on the manometer) can be found from the velocity equation of the venturi meter:

$$V_2 = C_v \sqrt{\frac{2(P_1 - P_2) / \rho_{air}}{\left(1 - \left(\frac{A_2}{A_1}\right)^2\right)}}$$
$$(P_1 - P_2) = \frac{\rho_{air} V_2^2 \left(1 - \left(\frac{A_2}{A_1}\right)^2\right)}{2C_v^2}$$

Since we assumed air is incompressible flow in this problem, then its density will stay constant at what was calculated above.

We need to find the  $C_v$  factor, in which Reynolds number needs to be calculated. Viscosity of air at 68 °F = 20 °C is found to be about  $1.9 \times 10^{-5}$  Pa.s.



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$$Re = \frac{\rho V_1 D_1}{\mu} = \frac{\rho \left( \frac{V_2 A_2}{A_1} \right) D_1}{\mu} = \frac{(4.9 \frac{kg}{m^3})(100m/s) \left( \frac{3inch}{6inch} \right)^2 (6inch \frac{1m}{39.37inc})}{1.9 \times 10^{-5} Pa \cdot s} = 9.8 \times 10^5$$

From charts of  $C_v$  vs. Reynolds number, you can find the  $C_v$  to be about 0.985.

$$(P_1 - P_2) = \frac{(4.9 \frac{kg}{m^3})(100 \frac{m}{s})^2 \left( 1 - \left( \frac{3inch}{6inch} \right)^4 \right)}{2(0.985)^2} = 23673.63 Pa = 23.7 kPa$$

Therefore, the manometer reading ( $\Delta h$ ) will be :

$$\Delta P = \rho_{Hg} g \Delta h$$

$$\Delta h = \frac{\Delta P}{\rho_{Hg} g} = \frac{23673.63 Pa}{(13600 kg / m^3)(9.8 m / s^2)} = 0.177 m$$