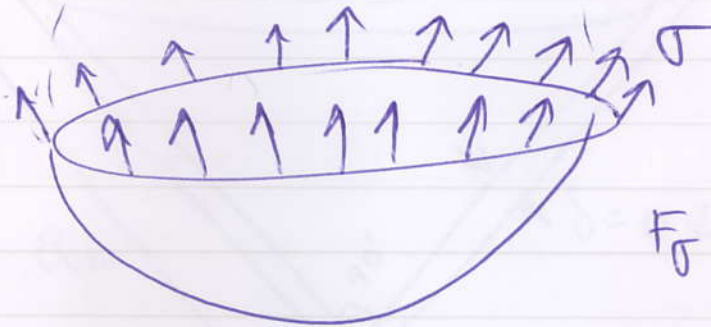


Problem 1 (10 marks)

→ DROPLET



$$\sum F = F_{\sigma} - F_p = 0$$

$$F_{\sigma} = \sigma L = \sigma(2\pi r)$$

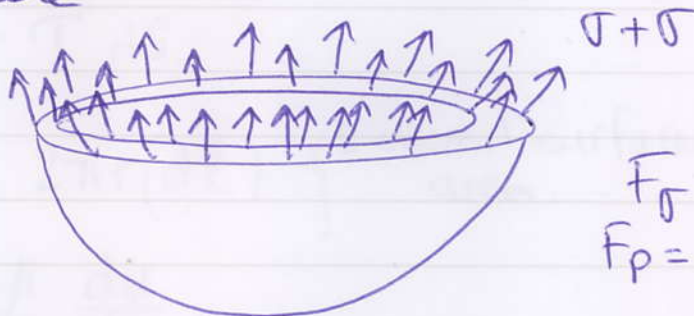
$$F_p = pA = p(\pi r^2)$$

$$\sigma(2\pi r) - p(\pi r^2) = 0 \Rightarrow \sigma(2\pi r) = p(\pi r^2)$$

$$p = \frac{\sigma(2\pi r)}{\pi r^2} = \frac{2\sigma}{r}$$

$$p = \frac{2(0.073 \text{ [N/m]})}{\frac{10}{2} \times 10^{-6} \text{ [m]}} = 29,200 \text{ Pa} \approx 29.2 \text{ kPa}$$

→ Bubble



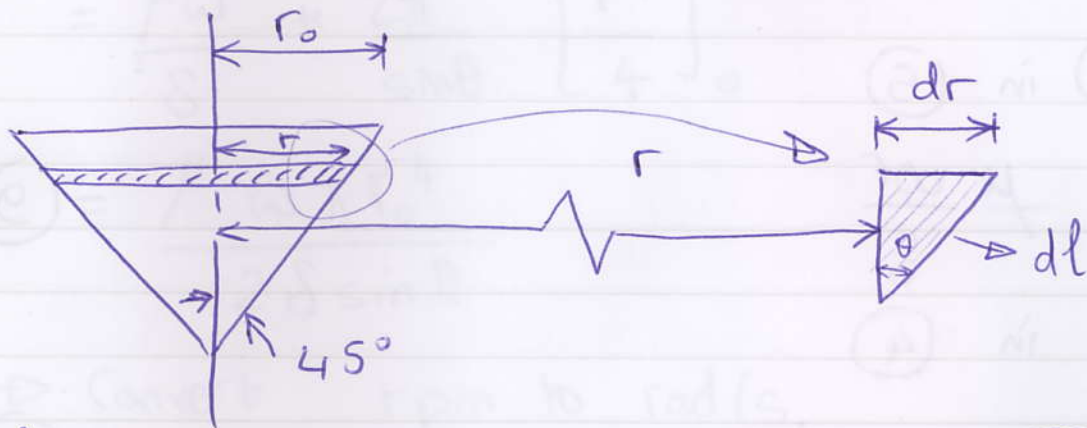
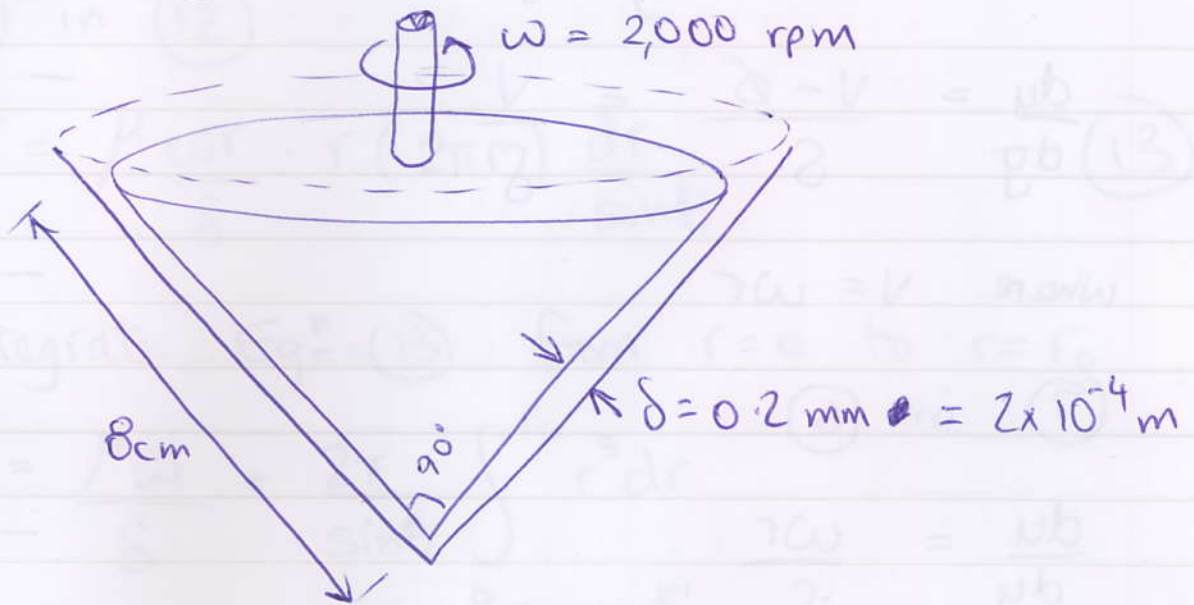
$$\sum F = F_{\sigma} - F_p = 0$$

$$F_{\sigma} = \sigma L = 2\sigma(2\pi r)$$

$$F_p = pA = p(\pi r^2)$$

$$p = \frac{4\sigma \pi r}{\pi r^2} = \frac{4\sigma}{r} = 58,400 \text{ Pa} \approx 58.4 \text{ kPa}$$

Problem 2 (15 MARKS)



$$dl = \frac{dr}{\sin \theta} \quad \text{--- (1)}$$

$$dT = r F_{\tau} \quad \text{--- (2)}$$

$$F_{\tau} = \tau ds \quad \text{--- (3)}$$

$$ds = 2\pi r(dl) \quad \left. \begin{array}{l} \text{Elemental surface} \\ \text{area} \end{array} \right\} \quad \text{--- (4)}$$

$$\tau = \mu \frac{du}{dy} \quad \text{--- (5)}$$

For a linear velocity profile

$$\frac{du}{dy} = \frac{V - \phi}{\delta} = \frac{V}{\delta} \quad \text{--- (6)}$$

where $V = \omega r$

(7) in (6)

$$\frac{du}{dy} = \frac{\omega r}{\delta} \quad \text{--- (7)}$$

(8) in (5)

$$\tau = \mu \frac{\omega r}{\delta} \quad \text{--- (8)}$$

(9) in (4)

$$dS = 2\pi r \frac{dr}{\sin\theta} \quad \text{--- (9)}$$

(10) in (3)

$$F_{\tau} = \tau \cdot 2\pi r \frac{dr}{\sin\theta} \quad \text{--- (10)}$$

(11) in (2)

$$dT = r \cdot 2\pi r \cdot \tau \cdot \frac{dr}{\sin\theta} \quad \text{--- (11)}$$

(12)

(9) in (12)

$$dT = \frac{\mu \omega r}{\delta} \cdot r (2\pi r) \frac{dr}{\sin\theta} \quad - (13)$$

Integrate Eqⁿ (13) from $r=0$ to $r=r_0$

$$T = \frac{\mu \omega}{\delta} * \frac{2\pi}{\sin\theta} \int_0^{r_0} r^3 dr$$

$$= \frac{\mu \omega}{\delta} * \frac{2\pi}{\sin\theta} \left[\frac{r^4}{4} \right]_0^{r_0}$$

$$= \frac{\mu \omega \pi r_0^4}{2 \delta \sin\theta} \quad - (14)$$

→ Convert rpm to rad/s

$$\omega = 2,000 \frac{\text{rev}}{\text{min}} * \frac{1 \text{ min}}{60 \text{ sec.}} * \frac{2\pi \text{ radians}}{\text{rev.}}$$

$$= 209.44 \text{ rad/sec.}$$

$$\text{Given } \nu_{\text{SAE-30}} = 1.1 \times 10^{-4} \text{ m}^2/\text{s}$$

$$\mu_{\text{SAE-30}} = \nu \rho_{\text{SAE-30}} \quad - (15)$$

$$\rho_{\text{SAE-30}} = 0.88 \quad (\text{Given})$$

$$\mu_{\text{SAE-30}} = \nu_{\text{SAE-30}} * \rho_{\text{SAE-30}} * \rho_{\text{water}}$$

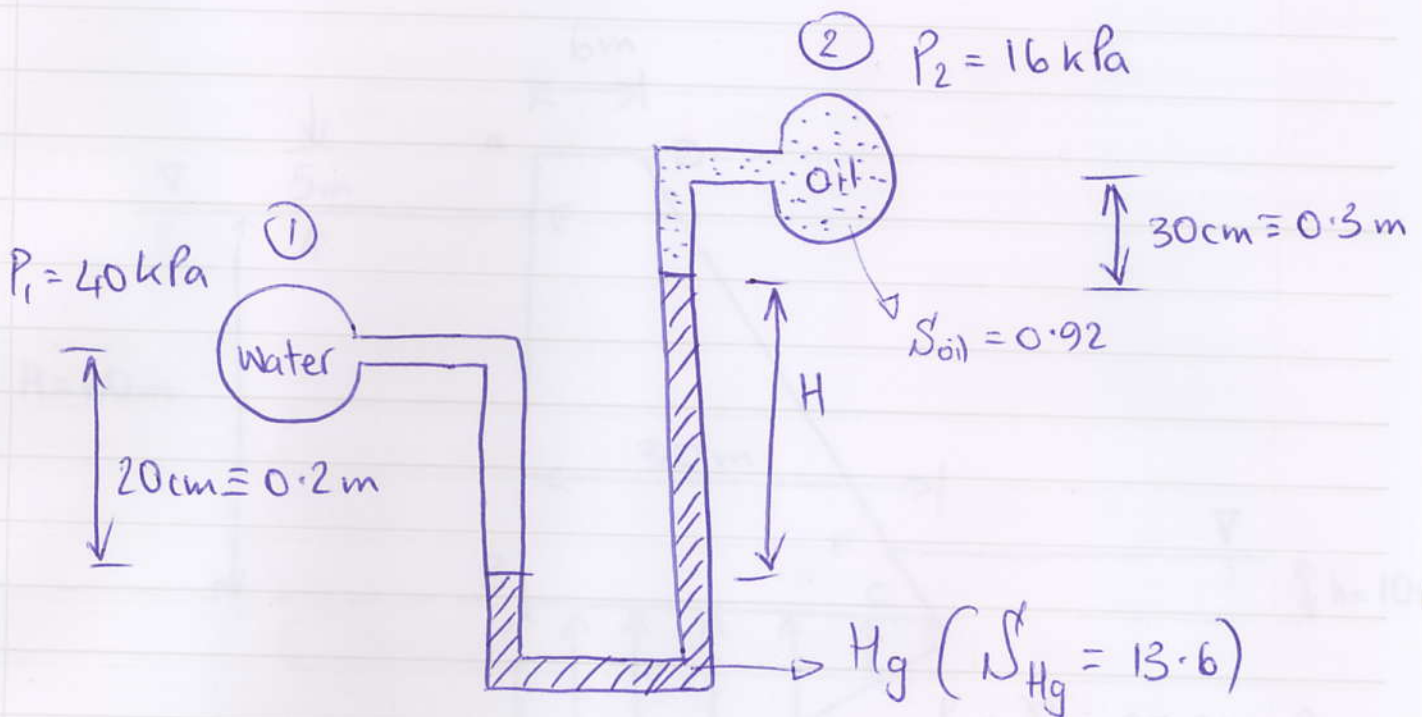
$$\begin{aligned} \mu_{\text{SAE-30}} &= 1.1 \times 10^{-4} \left[\frac{\text{m}^2}{\text{s}} \right] * 0.88 * 999 \left[\frac{\text{kg}}{\text{m}^3} \right] \\ &= 9.66 \times 10^{-2} \left[\frac{\text{Ns}}{\text{m}^2} \right] \end{aligned}$$

$$r_0 = 8 \text{ cm} \sin 45 = 5.66 \text{ cm} = 0.0566 \text{ m}$$

Eq. (15)

$$\begin{aligned} T &= \frac{9.66 \times 10^{-2} * 209.44 * \pi * 0.0566^4}{2 * 2 \times 10^{-4} * \sin 45} \\ &= 2.31 \text{ [Nm]} \end{aligned}$$

PROBLEM 3 (10 MARKS)



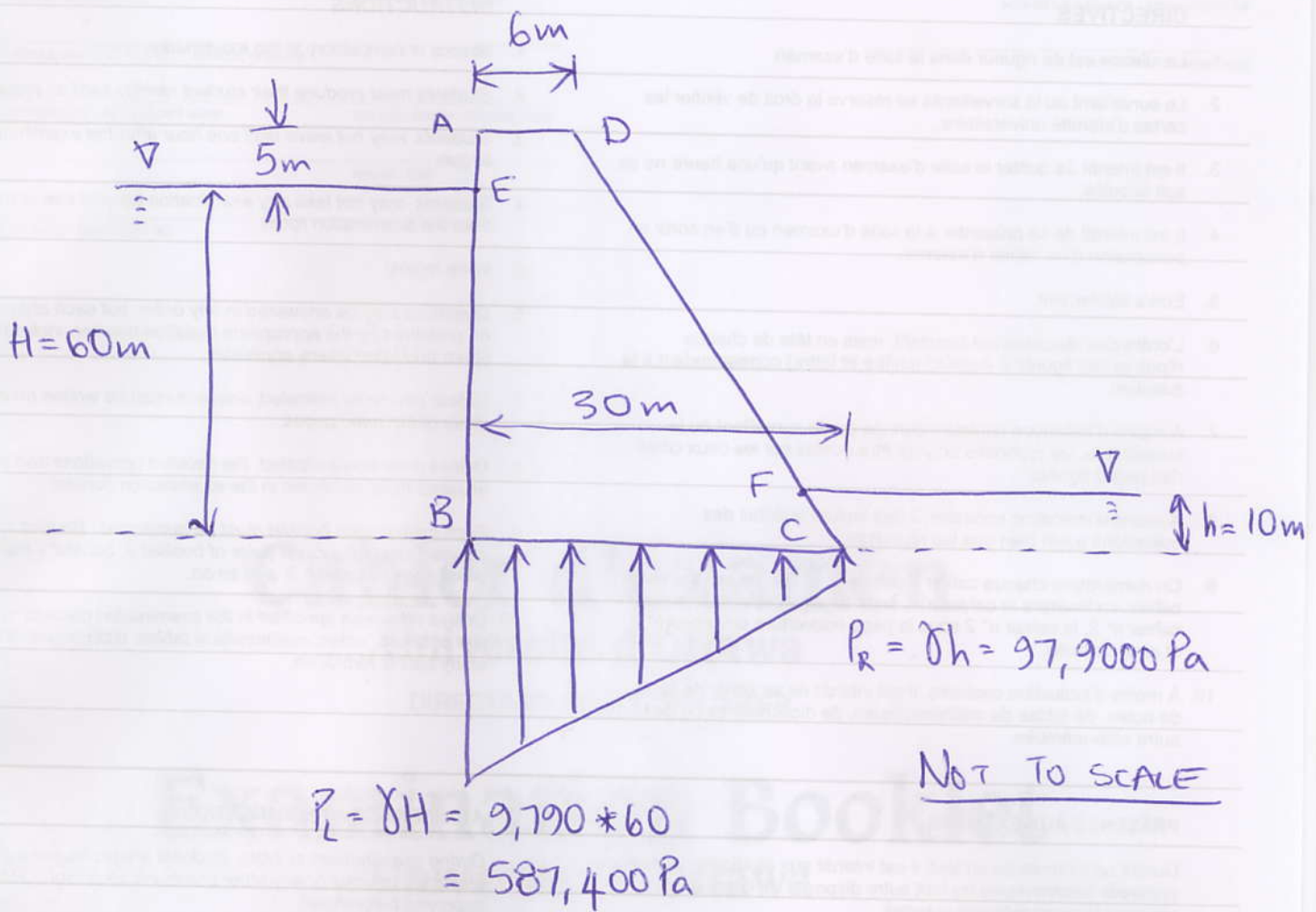
$$P_1 = P_2 + \gamma_{oil}(0.3) + \gamma_{Hg}(H) - \gamma_{water}(0.2)$$

$$= P_2 + \gamma_w S_{oil}(0.3) + \gamma_w S_{Hg}(H) - \gamma_w(0.2)$$

$$40000 = 16000 + 9790 [0.92 * 0.3 + 13.6H - 0.2]$$

$$H = 0.1747 \text{ m} \approx 17.5 \text{ cm}$$

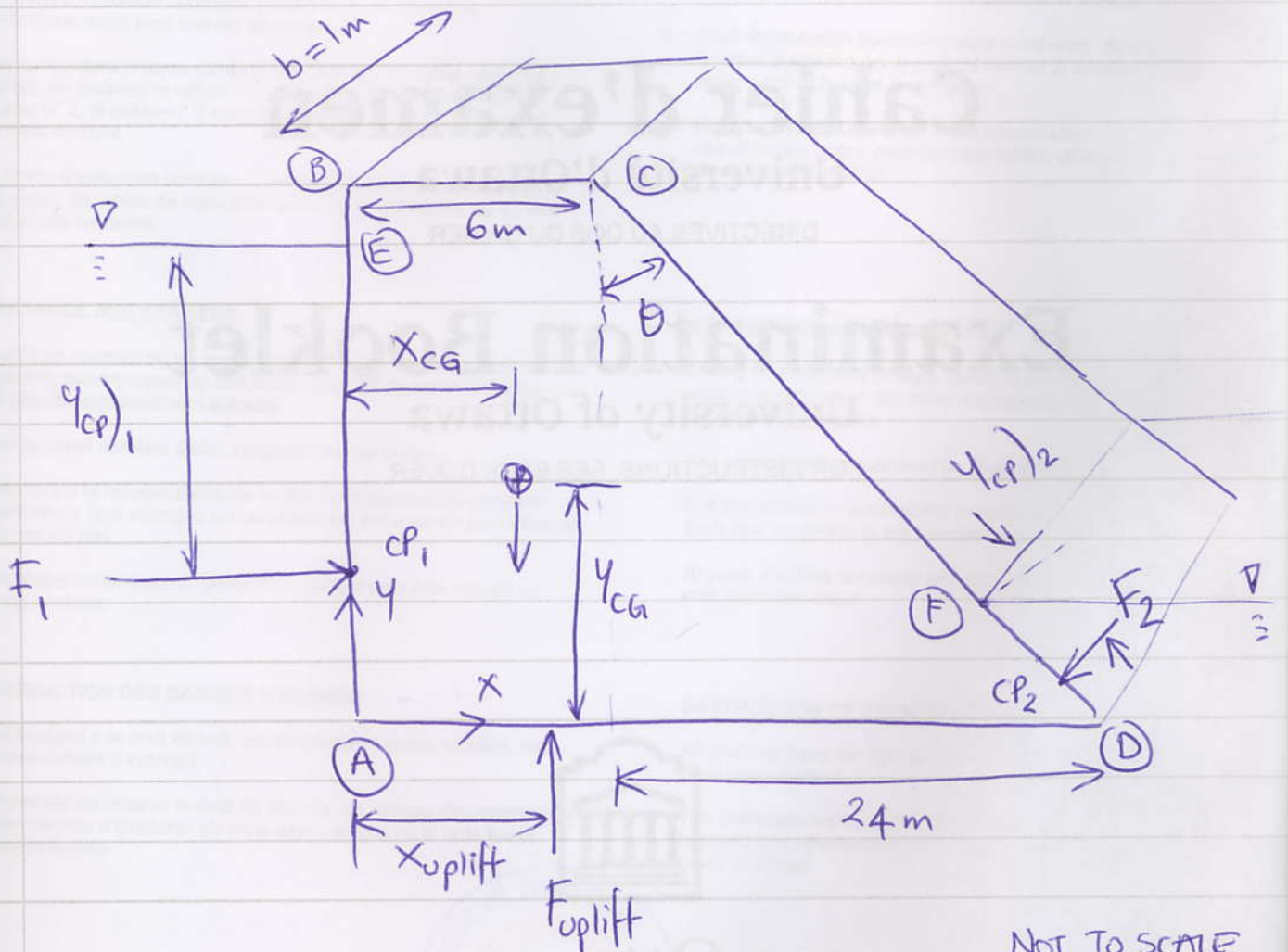
Problem 4 (15 MARKS)



Four forces acting on the structure

- 2 Hydrostatic forces (1 on the upstream (i.e. left) side & 1 on the downstream (i.e. right side))
- 1 Uplift force (which is really a hydrostatic force present in the soil/ground)
- 1 weight of the structure

All the forces mentioned above must be analyzed as to how will they influence stability (overturning) of the structure. It seems that the uplift & left hydrostatic force will cause the structure to turn clockwise while the right hydrostatic force & weight would cause the structure to turn counter-clockwise about the right-hand corner. However, this overturning moment must be quantified.



① ~~□~~ F_1

$$F_1 = \bar{P}_1 A_1$$

$$= \gamma \bar{h}_1 A_1$$

$$= \gamma \frac{H}{2} b H$$

$$= \frac{\gamma H^2 b}{2}$$

$$= \frac{9,790 (60)^2 b}{2} = 17.62 b \text{ [MN]}$$

$$y_{cp,1} = \bar{y}_1 + \frac{\bar{I}_1}{\bar{y}_1 A_1}$$

$$= \frac{H_1}{2} + \frac{b H_1^3}{12 (H_1/2) b H_1} = \frac{2H}{3} = 40 \text{ [m]}$$

$\bar{h}_1 =$ height to centroid

$$= H_1/2$$

$$A_1 = b H_1$$

$$I_1 = b H_1^3 / 12$$

$$\bar{y}_1 = H_1/2$$

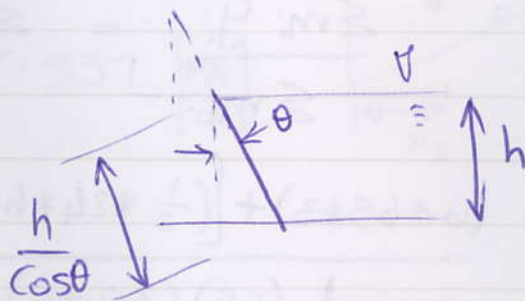
② F_2

$$F_2 = \bar{P}_2 A_2$$

$$= \gamma \bar{h}_2 A_2$$

$$= \gamma \frac{h}{2} \frac{bh}{\cos \theta}$$

$$= \frac{\gamma b h^2}{2 \cos \theta} = 0.522 b \text{ [MN]}$$



$\bar{h}_2 =$ height to centroid

$$= h/2$$

$$A_2 = bh / \cos \theta$$

$$\theta = \tan^{-1} \frac{24}{65} = 20.27^\circ$$

$$Y_{cp} = \bar{y}_2 + \frac{\bar{I}_2}{\bar{y}_2 A_2}$$

$$= \frac{h}{2 \cos \theta} + \frac{b(h/\cos \theta)^3}{12(h/\cos \theta) b(h/\cos \theta) * \frac{1}{2}}$$

$$\cos \theta = 0.938$$

$$\bar{y}_2 = h/2 \cos \theta$$

$$\bar{I}_2 = b(h/\cos \theta)^3 / 12$$

$$= \frac{2h}{3 \cos \theta}$$

$$= 7.11 \text{ m}$$

(3) W

$$W = \gamma_{\text{concrete}} V_{\text{concrete}} = \gamma_w N_{\text{concrete}} V_{\text{concrete}}$$

$$V_{\text{concrete}} = \frac{1}{2} 65(6+30) * b = ~~1170~~ 1,170b$$

$$W = 9790 * 2.4 * 1,170b = 27.49b \text{ [MN]}$$

$$X_{CG} = \frac{\sum m_i X_i}{\sum m_i} = \frac{\sum W_i X_i}{\sum W_i} = \frac{\sum A_i X_i}{\sum A_i}$$

$$Y_{CG} = \frac{\sum m_i Y_i}{\sum m_i} = \frac{\sum W_i Y_i}{\sum W_i} = \frac{\sum A_i Y_i}{\sum A_i}$$

$$X_{CG} = \frac{(6 * 65 * 3) + \left[\left(\frac{1}{2} * 24 * 65 \right) \left(6 + \frac{24}{3} \right) \right]}{\frac{1}{2} (65) (30 + 6)} = \frac{1,170 + 10,920}{1,170}$$

$$= 10.33 \text{ m}$$

$$Y_{CG} = \frac{(6 * 65 * \frac{65}{2}) + \left[\left(\frac{1}{2} * 24 * 65 \right) * \left(\frac{65}{3} \right) \right]}{\frac{1}{2} (65) (30 + 6)} = \frac{12,675 + 16,900}{1,170}$$

$$= 25.28 \text{ m}$$

~~$$y_{\text{cep}} = y_2 + \frac{I_2}{y_2 A_2}$$~~

④ F_{uplift}

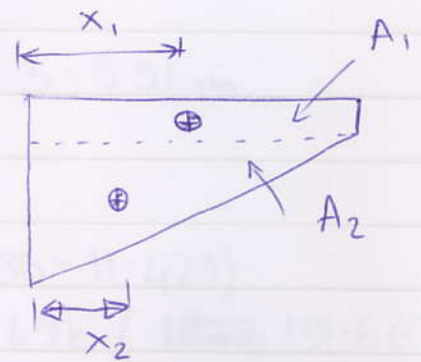
$$F_{\text{uplift}} = P_{\text{uplift-resultant}} * A_{\text{bottom}}$$

$$A_{\text{bottom}} = 30 * b = 30b \text{ [m]}$$

$$P_{\text{uplift-resultant}} = \frac{P_L + P_R}{2} = \frac{587,400 + 97,900}{2} = 342,650 \text{ Pa}$$

$$F_{\text{uplift}} = 342,650 * 30b = 10.28 \text{ [MN]}$$

$$X_{\text{uplift}} = \frac{A_1 x_1 + A_2 x_2}{\Sigma A}$$



$$A_1 = 97,900 * 30 = 2.937 \left[\frac{\text{MN}}{\text{m}} \right]$$

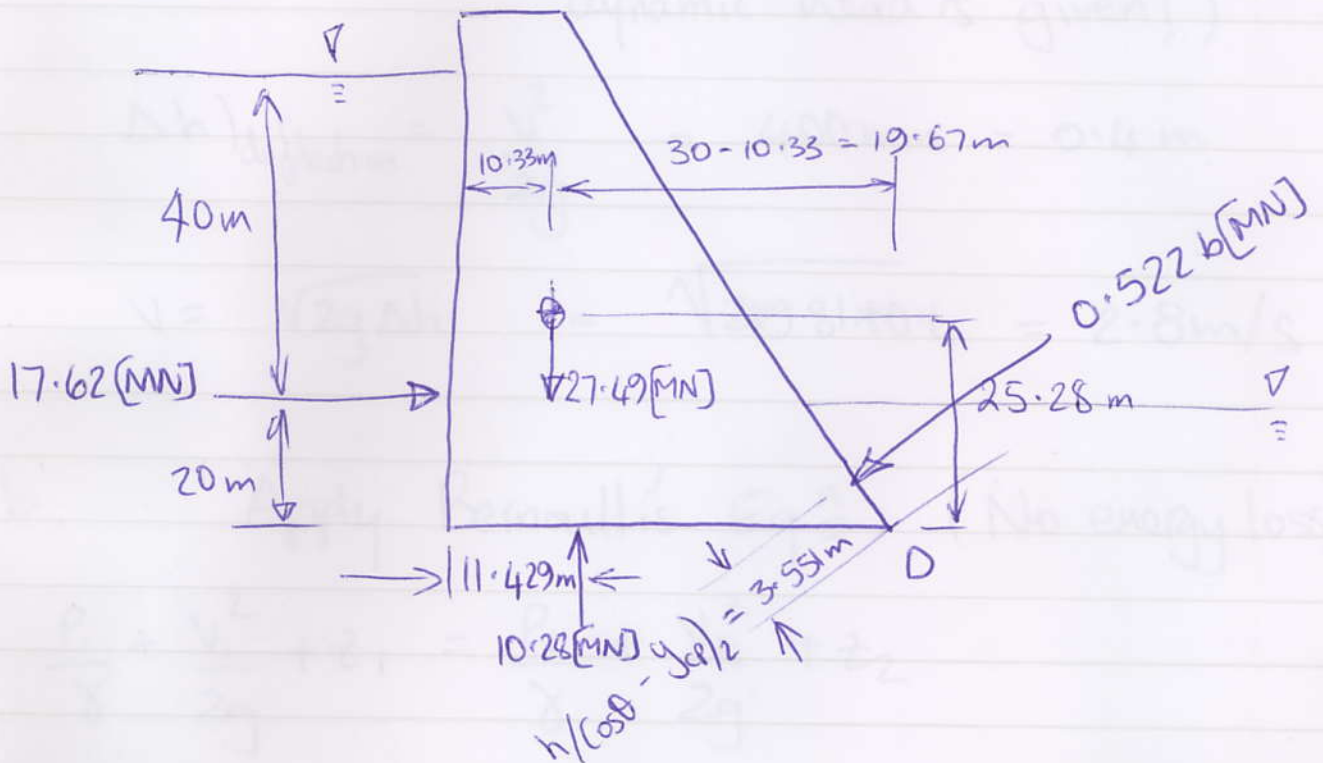
$$x_1 = 15 \text{ m}$$

$$A_2 = (587,400 - 97,900) * \frac{1}{2} * 30 = 7.343 \left[\frac{\text{MN}}{\text{m}} \right]$$

$$x_2 = 30/3 = 10 \text{ m}$$

$$X_{\text{uplift}} = \frac{2.937 * 15 + 7.343 * 10}{2.937 + 7.343} = 11.429 \text{ [m]}$$

→ Summation of Moments



$$\frac{h}{\cos\theta} - y_{cp} = \frac{10}{0.938} - 7.11 \text{ m} = 3.551 \text{ m}$$

$$\begin{aligned} \sum M_D &= +17.62b(20) + 10.28b(30 - 11.429) \\ &\quad - 0.522b(3.551) - 27.49b(\text{---} 19.667) \\ &= 0.81 \text{ (Nm)} \rightarrow +ve \therefore \text{will overturn about D} \end{aligned}$$

Problem 5 (10 Marks)

a. Δh deflection \rightarrow dynamic head (i.e. no need to consider static head, since the dynamic head is given!)

$$\Delta h)_{\text{deflection}} = \frac{V^2}{2g} = 400 \text{ mm} = 0.4 \text{ m}$$

$$V = \sqrt{2g\Delta h} = \sqrt{2 \times 9.81 \times 0.4} = 2.8 \text{ m/s}$$

b. Apply Bernoulli's Eqⁿ (No energy loss)

$$\frac{P_1}{\gamma} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\gamma} + \frac{V_2^2}{2g} + z_2$$

$$z_1 = z_2 \quad (\text{same datum})$$

$$P_1/\gamma = 0.25 \text{ m}$$

$$P_2/\gamma = H$$

$$0.25 + \frac{0.5^2}{2(9.81)} = H + \frac{1.125^2}{2(9.81)}$$

$$H = 0.198 \text{ m}$$

Problem 6 (10 MARKS)

CONTINUITY EQN

$$\frac{d}{dt} \int_{cv} \rho dV + \int_{cs} \rho \vec{v} \cdot d\vec{A} = 0$$

$$\frac{dM_{cv}}{dt} = \sum \dot{m}_{in} - \sum \dot{m}_{out}$$

$$M_{cv} = \rho V_{cv}$$

$$\frac{dM_{cv}}{dt} = \frac{d}{dt} (\rho V_{cv}) = \frac{d}{dt} (\rho A_{cv} h_{cv}) = \rho A_{cv} \frac{dh_{cv}}{dt}$$

$$\dot{m}_{in}|_1 = \rho A_1 v_1 = 998 * \frac{\pi}{4} (0.04)^2 * 10 = 12.54 \text{ kg/s}$$

$$\dot{m}_{in}|_2 = \rho A_2 v_2 = \rho Q_2 = \dot{m}_2 = 10 \text{ kg/s}$$

$$\sum \dot{m}_{in} = 12.54 + 10 = 22.54 \text{ kg/s} \quad \left[\frac{1 \text{ min}}{60 \text{ sec}} \right] \cdot \left[\frac{\text{m}^3}{1000 \text{ L}} \right]$$

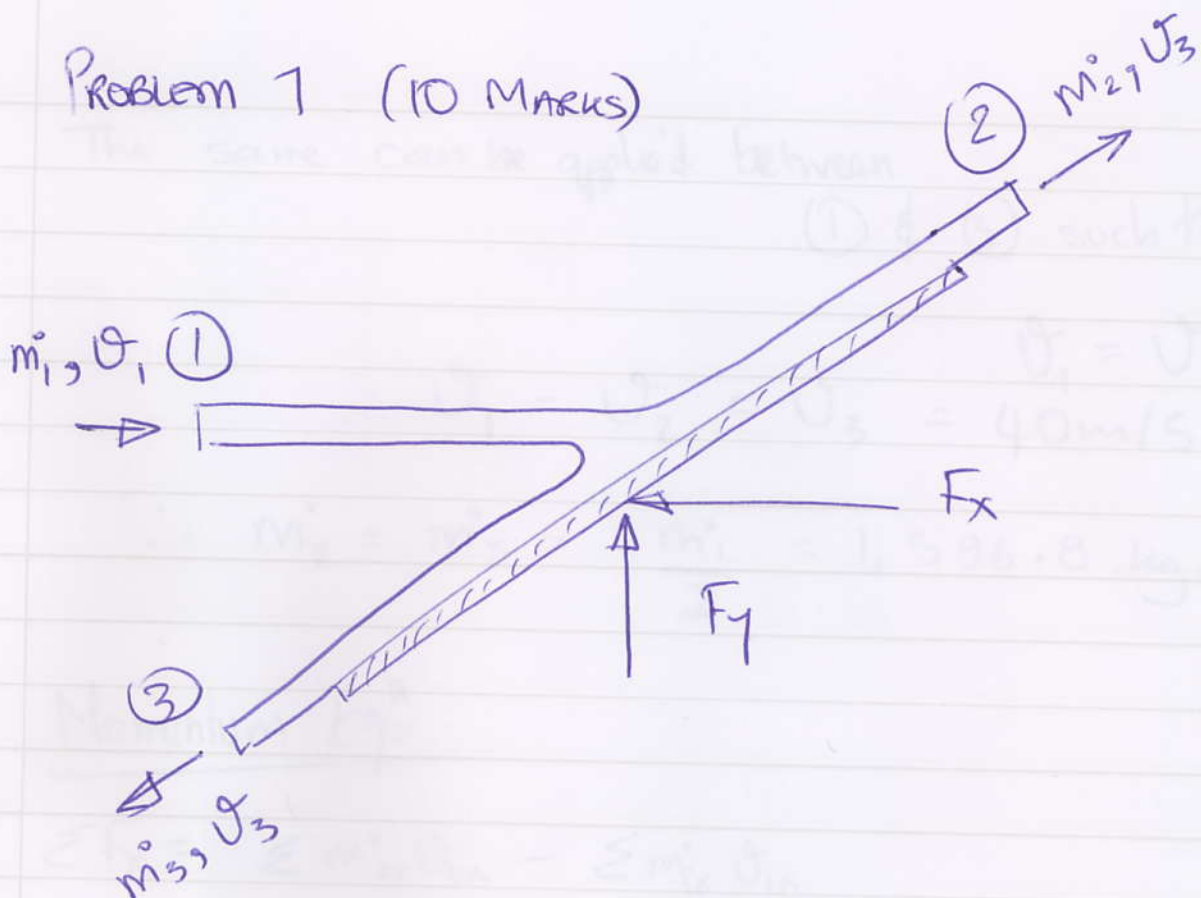
$$\sum \dot{m}_{out} = \dot{m}_3 = \rho Q_3 = 998 \left[\frac{\text{kg}}{\text{m}^3} \right] * 600 \left[\frac{\text{L}}{\text{min}} \right] * (\text{conversion})$$
$$= 9.98 \text{ kg/s}$$

$$\frac{dh_{cv}}{dt} = \frac{\sum \dot{m}_{in} - \sum \dot{m}_{out}}{\rho A_{cv}} = \frac{22.54 - 9.98}{998 * \pi (1.2^2)/4}$$

$$= 0.0111 \text{ [m/s]}$$

$$= 1.11 \text{ [cm/s]}$$

Problem 1 (10 MARKS)



Continuity

$$\dot{m}_1 = \dot{m}_2 + \dot{m}_3$$

$$\begin{aligned} \dot{m}_1 &= \rho U_1 A_1 = \rho U_1 (h \cdot b) \\ &= 998 \cdot 40 \text{ m/s} \cdot 0.2 \cdot 0.4 \\ &= 3193.6 \text{ kg/s} \end{aligned}$$

Bernoulli's Eqⁿ

$$\frac{P_1}{\gamma} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\gamma} + \frac{V_2^2}{2g} + z_2$$

$$\begin{aligned} z_1 &= z_2 \\ P_1 &= P_2 = 0 \end{aligned}$$

$$\therefore U_1 = U_2$$

The same can be applied between

① & ③ such that

$$\therefore V_1 = V_2 = V_3 = 40 \text{ m/s} \quad V_1 = V_3$$

$$\therefore \dot{m}_2 = \dot{m}_3 = \frac{\dot{m}_1}{2} = 1,596.8 \text{ kg/s}$$

Momentum Eqⁿ

$$\sum F_x = \sum \dot{m}_{ox} V_{ox} - \sum \dot{m}_{ix} V_{ix}$$

$$-F_x = \left[-\dot{m}_3 * 40 \cos 45 + \dot{m}_2 * 40 \cos 45 \right] - \dot{m}_1 * 40$$

$$-F_x = -\dot{m}_1 * 40 = -127,744 \text{ [N]}$$

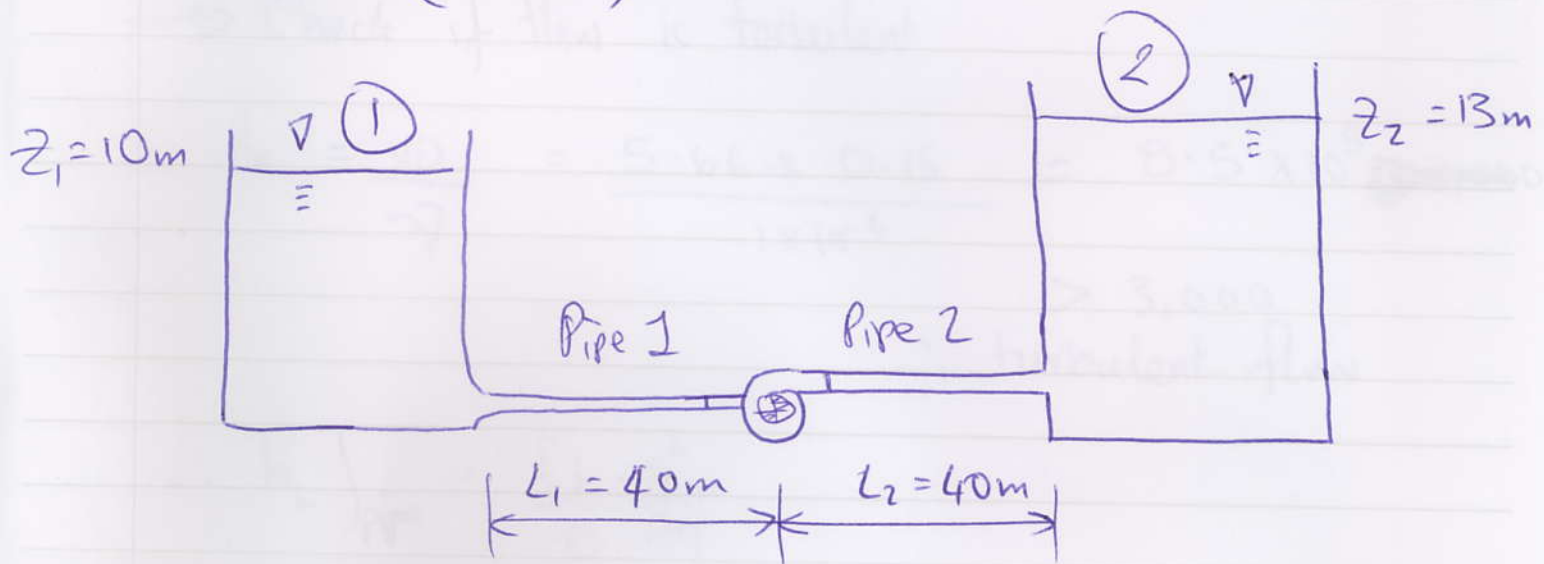
$$\therefore F_x = 127.74 \text{ [kN]}$$

$$\sum F_y = \sum \dot{m}_{oy} V_{oy} - \sum \dot{m}_{iy} V_{iy}$$

$$F_y = \left[-\dot{m}_3 * 40 \sin 45 + \dot{m}_2 * 40 \sin 45 \right] - [0]$$

$$F_y = 0 \text{ [N]}$$

Problem 8 (15 MARKS)



$$Q = 0.10 \text{ m}^3/\text{s}$$

$$D_1 = D_2 = 0.15 \text{ m}$$

$$V_{P_1} = V_{P_2} = \frac{Q}{A}$$

(where "p" stands for "pipe"!)

$$V_{\text{pipe 1}} = \frac{0.10}{\pi (0.15)^2 / 4}$$

$$= 5.66 \text{ m/s}$$

$$= V_{\text{pipe 2}}$$

Energy Equation

$$z_1 + \frac{P_1}{\gamma} + \alpha_1 \frac{V_1^2}{2g} + h_p = z_2 + \frac{P_2}{\gamma} + \alpha_2 \frac{V_2^2}{2g} + h_t + \sum h_L$$

(atm.)
(negl.)
(atm.)
(negl.)
(no turbine)

* NOTE: $V_1 \neq V_{\text{pipe 1}}$ & $V_2 \neq V_{\text{pipe 2}}$

V_1 = vel. of water surface at (1)
 V_2 = vel. of water surface at (2)

$$\therefore h_p = [z_2 - z_1] + \sum h_L$$

→ Check if flow is turbulent

$$Re = \frac{VD}{\nu} = \frac{5.66 \times 0.15}{1 \times 10^{-6}} = 8.5 \times 10^5$$

$> 3,000$
 \therefore turbulent flow

$$\therefore h_L \Big|_{\text{pipe}} = \frac{fL}{D} \frac{V_{\text{pipe}}^2}{2g}$$

$$\begin{aligned} \sum h_L &= \frac{f_1 L_1}{D_1} \frac{V_{\text{pipe}1}^2}{2g} + \frac{f_2 L_2}{D_2} \frac{V_{\text{pipe}2}^2}{2g} \\ &= \frac{f(L_1 + L_2)}{D} \frac{V_p^2}{2g} \end{aligned}$$

$$f = 0.015 \quad ; \quad L = 40 + 40 = 80 \text{ m} \quad ; \quad D = 0.15 \text{ m}$$

$$\sum h_L = \frac{0.015(80)}{0.15} \times \frac{5.66^2}{2(9.81)} = 13.192 \text{ m}$$

$$\begin{aligned} h_p &= (z_2 - z_1) + \sum h_L = 13 - 10 + 13.192 \text{ m} \\ &= 16.192 \text{ [m]} \end{aligned}$$

$$\begin{aligned} P_{\text{pump}} &= \gamma Q h_p = 9,790 \times 0.1 \times 16.192 \\ &= 15,851.95 \text{ W} \\ &= 15.85 \text{ kW} \end{aligned}$$

BONUS QUESTION (5 MARKS)

$$k_s = 0.046 \text{ mm} = 0.046 \times 10^{-3} \text{ m}$$

$$\frac{k_s}{D} = \frac{0.046 \times 10^{-3}}{0.15} = 0.0003$$

$$Re = 8.5 \times 10^5 \text{ (previously calculated)}$$

Using $k_s/D = 0.0003$ & $Re = 8.5 \times 10^5$ in Moody's Diagram

$\rightarrow f \approx 0.016$ (hence, it is not correct to assume $f = 0.015$)

If Q is unknown, f is unknown & k_s is known, and h_p is known, then k_s/D can be calculated, but without, Q , U_p cannot be calculated & Reynold's number, Re , cannot be obtained either. However, the following steps can be made to obtain a value for U_p :

- 1- Assume fully turbulent conditions (straight line region in Moody's diagram)
 \hookrightarrow get "f" from k_s/D diagram
- 2- From energy eqⁿ get U_p (h_p , U_p & Δz & Δh_L known)
- 3- Calculate Re from U_p above.
- 4- Obtain new value for "f" using Moody diagram
- 5- Repeat steps 1-4 until "f" converges (does not change)

Problem 9

- a. No. PIPE AB is the smallest
- b. A
- c. C
- d. Jet
- e. Sudden Expansion

